

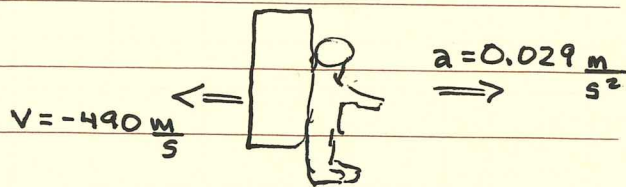
Chapter 8 Homework Problem

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| DATE | |
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Ex. 61

A 70-kg astronaut floating in space in a 110 kg MMU

$$M = 70 \text{ kg} + 110 \text{ kg} = 180 \text{ kg}$$



b.) What is the thrust?

$$\text{Thrust} = F = Ma = (180 \text{ kg})(0.029 \text{ m/s}^2) = \boxed{5.22 \text{ N}}$$

a.) How much gas is used? ($\Delta t = 5.0$ seconds)

$$\text{Thrust} = v_{\text{ex}} \frac{dm}{dt} \quad \frac{dm}{dt} = \frac{\text{Thrust}}{v_{\text{ex}}} = \frac{5.22 \text{ N}}{490 \text{ m/s}} = 0.01065 \frac{\text{kg}}{\text{sec}}$$

$$\Delta m = \text{mass of gas used in } (\Delta t = 5.0 \text{ sec}) = \frac{dm}{dt} \Delta t$$

$$\Delta m = 0.01065 \text{ kg/sec} (5.0 \text{ sec}) = 0.053 \text{ kg} \Rightarrow \boxed{53 \text{ grams}}$$

Ex. 82

This is a 2-part problem:

1.) Use conservation of momentum to find v_f after the "completely inelastic collision."

when it reaches

The velocity of m at the bottom is:

$$v = \sqrt{2gh} = \sqrt{2gR}$$

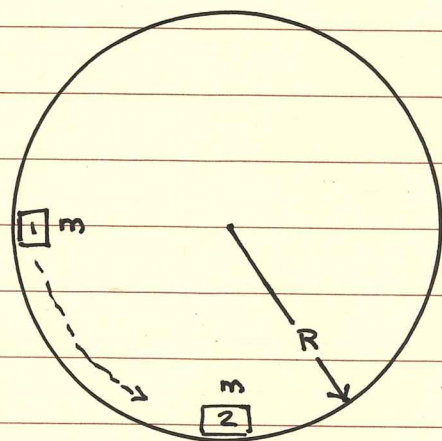
It strikes and "sticks" to the 2nd mass.

Cons. of Mom \Rightarrow

$$m_1 v_1 = (m_1 + m_2) v_f = m \sqrt{2gR} = 2m v_f$$

$$\boxed{v_f = \sqrt{\frac{gR}{2}}}$$

velocity at the bottom for both masses once they're "stuck" together.



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Ex. 82 (Cont'd)

2.) Use conservation of energy to see how high the 2 blocks rise.

$$W_{\text{TOT}} = \Delta K = K_f - K_i = -K_i$$

$$-mgy_f = -\frac{1}{2}mV_i^2 \quad \text{where } "m" = 2m$$

$$gy_f = \frac{1}{2}V_i^2 \quad y_f = \frac{V_i^2}{2g} = \frac{(\sqrt{gR/2})^2}{2g} = \frac{gR}{4g}$$

$$y_f = \frac{R}{4}$$

Ex. 95

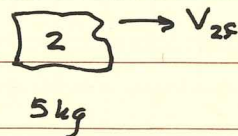
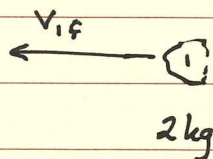
A 7.0-kg shell at rest explodes into two fragments

$V = 0$ (at rest)

Initial: \Rightarrow



Final: \Rightarrow



$K_1 = ?? \text{ J}$

$K_2 = 100 \text{ J}$

$$P_2 = \sqrt{2m_2K}$$

However, from cons. of momentum, $P_1 = P_2$

$$P_1 = P_2 = \sqrt{2m_2K_2}$$

$$K_1 = \frac{P_1^2}{2m_1} = \frac{2m_2K_2}{2m_1} = \frac{m_2}{m_1} K_2 = \frac{5 \text{ kg}}{2 \text{ kg}} (100 \text{ J}) = 250 \text{ J}$$

$$K_1 = 250 \text{ J}$$

Δv for Rocket Propulsion

$$\Delta v = v_{ex} \ln \left(\frac{M_i}{M_f} \right)$$

where $\Delta v = v_f - v_i$ v_{ex} = exhaust velocity w.r.t. the rocket

M_i = initial mass of the rocket

M_f = final mass of the rocket

8.63

If the rocket ejects gas at a relative speed of 1900 m/s

$$v_i = 0 \text{ m/s}$$

$$v_f = \frac{1}{1,000} \text{ (speed of light)} = 3.00 \times 10^5 \text{ m/s}$$

$$\ln \left(\frac{M_i}{M_f} \right) = \frac{\Delta v}{v_{ex}}$$

$$\ln \left(\frac{M_f}{M_i} \right) = -\frac{\Delta v}{v_{ex}} = -158$$

$$\left(\frac{M_f}{M_i} \right) = e^{-\frac{\Delta v}{v_{ex}}} = 2.67 \times 10^{-69}$$

$$\frac{M_f}{M_i} = 2.67 \times 10^{-69}$$

b.) what is the fraction if the final speed is 3200 m/s

$$\frac{M_f}{M_i} = e^{-\frac{\Delta v}{v_{ex}}} = e^{-\frac{3200 \text{ m/s}}{1900 \text{ m/s}}} = 0.186$$

$$\frac{M_f}{M_i} = 0.186$$